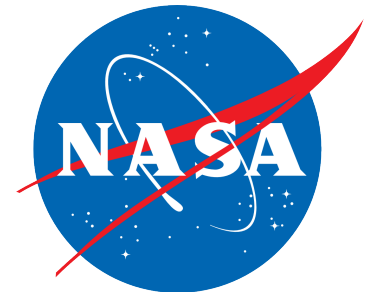




# Low-Energy Proton Single Event Effects

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1. Institute for Space and Defense Electronics
2. Vanderbilt University
3. NASA Goddard Space Flight Center
4. Applied Physics Laboratory
5. Texas Instruments



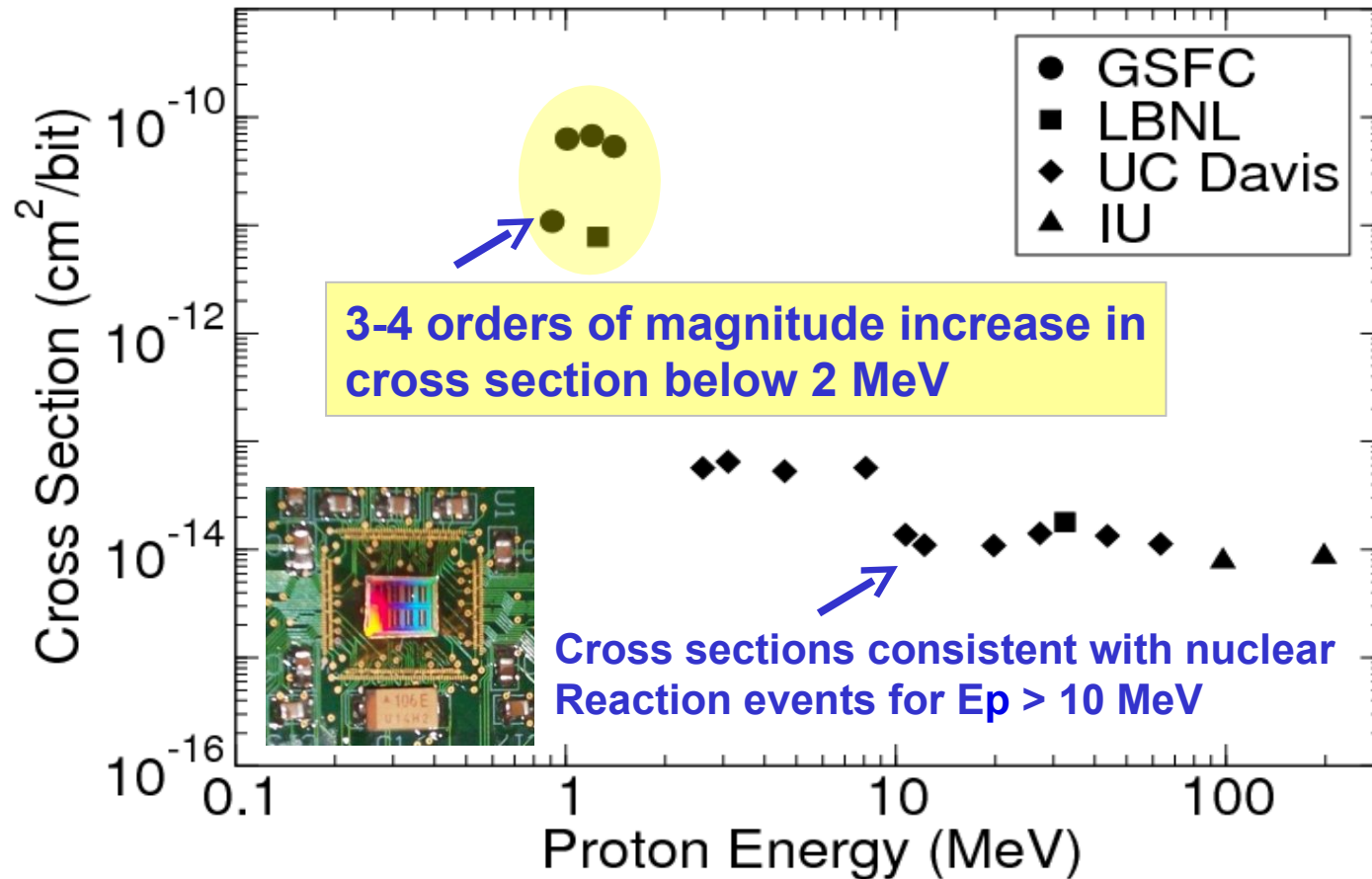


# Low-Energy Proton Upset Cross Sections



Vanderbilt Engineering

Data collected by Vanderbilt and NASA Goddard on TI 65 nm bulk CMOS process



Consistent with evidence of proton direct ionization contributing to single event upsets reported for IBM 65 nm SOI process [Rodbell, TNS 2007][Heidel, TNS 2008]



# Key Contributions



Vanderbilt Engineering

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- **Creation of an upset model based on first principles and calibrated to well-controlled radiation sources to predict the low-energy proton response**
- **Application of the model to predict the response in the space environment**
- **First unified rate prediction of full space environment including direct ionization from protons**



# Device Under Test



Vanderbilt Engineering

- **High density 6-transistor SRAM from Texas Instruments 65 nm bulk CMOS process**
  - 4 Megabit
  - Chip-on-board
  - Operated at 1.2 Volts
- **FPGA-based tester and test board designed by NASA Goddard**

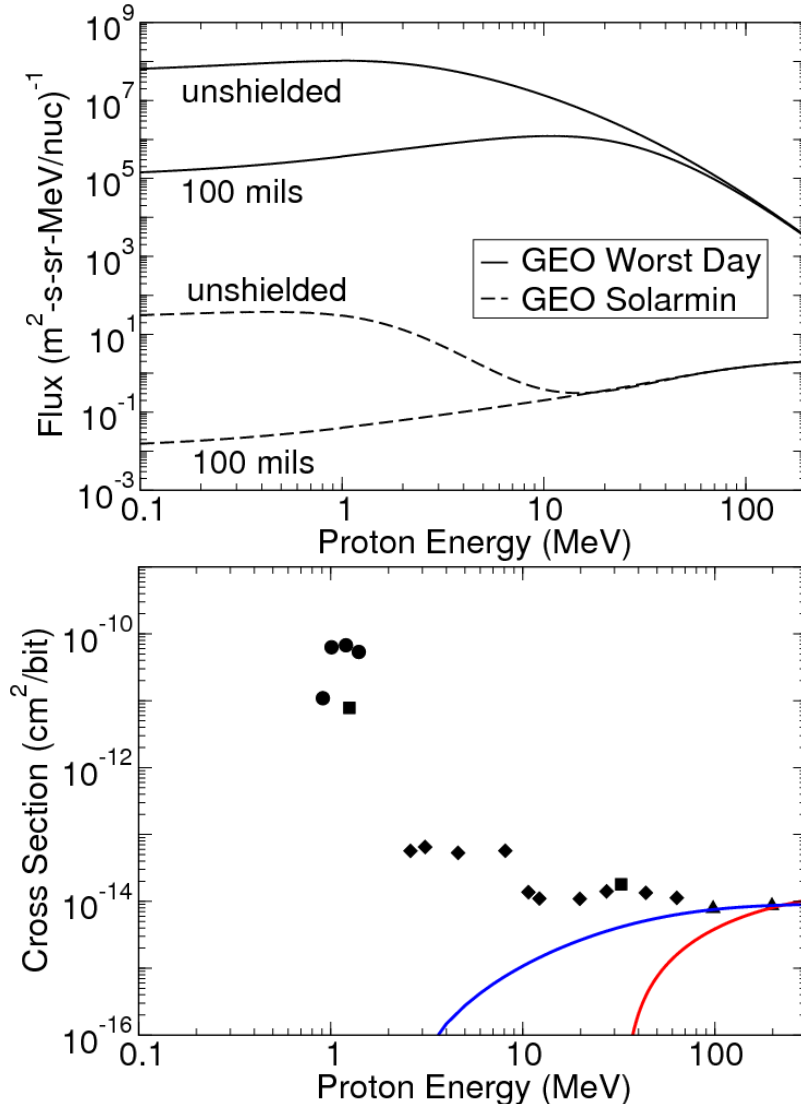




# Transported Proton Spectra



Vanderbilt Engineering



- **Proton flux that reaches device in CREME96 “Worst Day” environment is significant even with heavy shielding**
  - Low-energy protons not eliminated
- **Flux not significantly attenuated with heavier shielding in “Solar min”**
- **Bendel models provide relationship between kinetic energy and SEU cross section driven by nuclear reactions**
  - Neither adequately model the direction ionization mechanism

$$\sigma = \left(\frac{B}{A}\right)^{14} \left(1 - e^{(-0.18Y^{\frac{1}{2}})}\right)^4$$

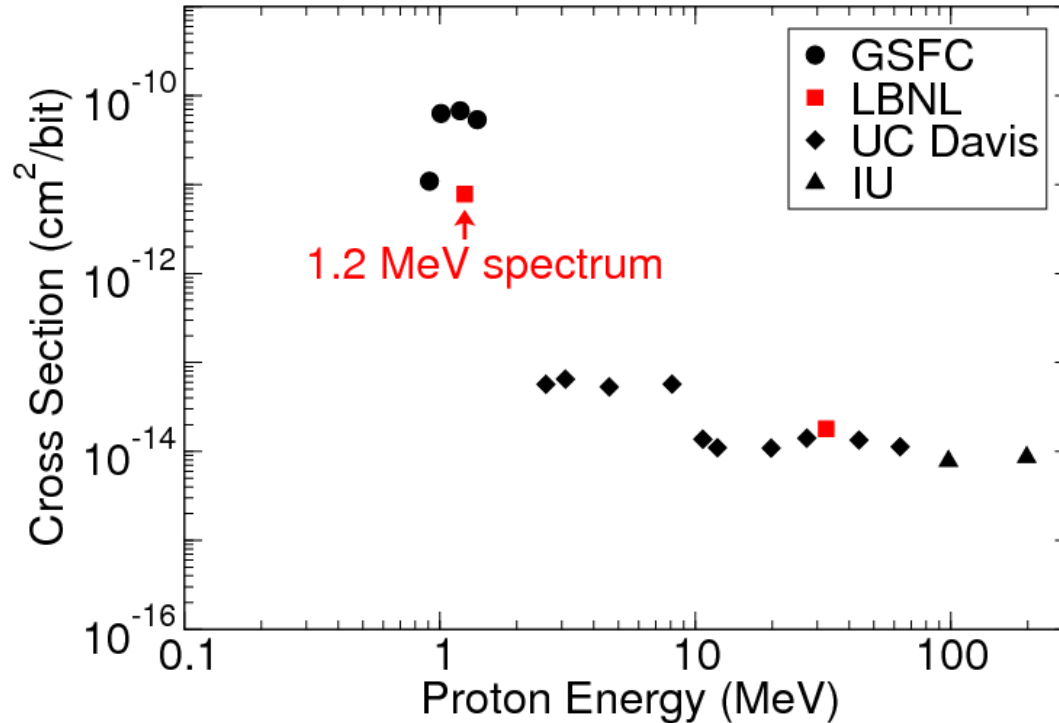
$$Y = \left(\frac{18}{A}\right)^{\frac{1}{2}} (E - A)$$



# Low-Energy Mechanism



Vanderbilt Engineering



- **Data obtained at LBNL provide proof that the increase in SEU cross section is a real effect**
  - Same test setup, beamline, and dosimetry
  - Increase in cross section > 400X
- **Low-energy test has large spread in energy, can be used to determine if the device has a proton sensitivity, but difficult to quantify**

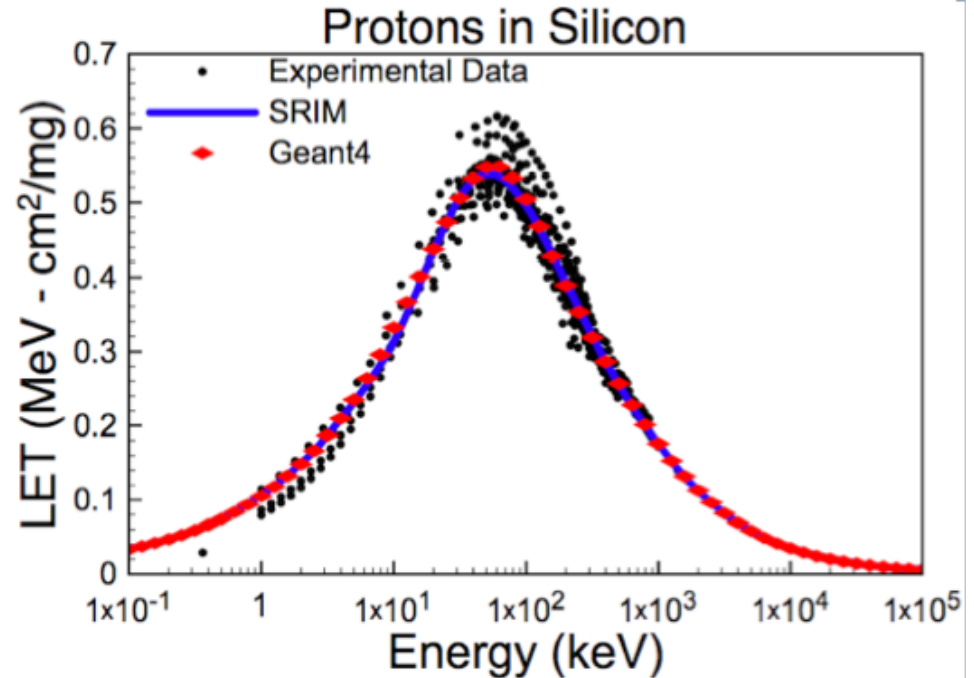


# Stopping Power Uncertainty



Vanderbilt Engineering

- **Low proton energy leads to several important topics**
  - Where's the Bragg peak?
  - Tune the beam or degrade it
  - Topside testing (wire-bonded DUT) or backside (C4)
  - Straggling, which affects both range and energy
- **Hard to fit an error rate prediction model with unknowns in incident particles**
  - Real physics is important and not well characterized by  $dE/dx$
  - Systematic complication from both an experimental and simulation perspective
- **SRIM cites variation in reported stopping power and Bragg peak**



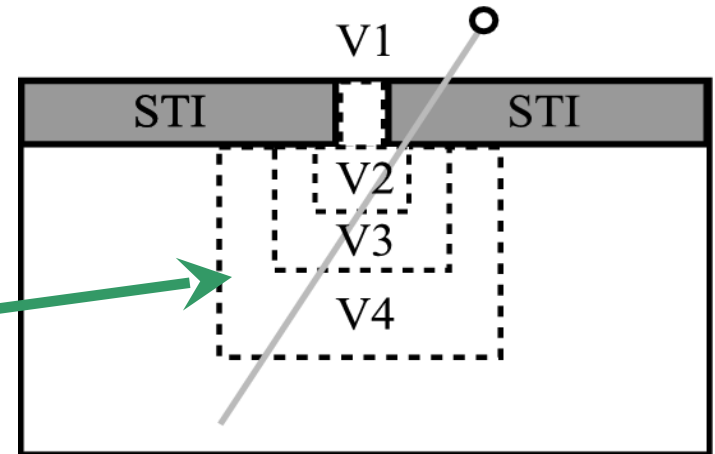
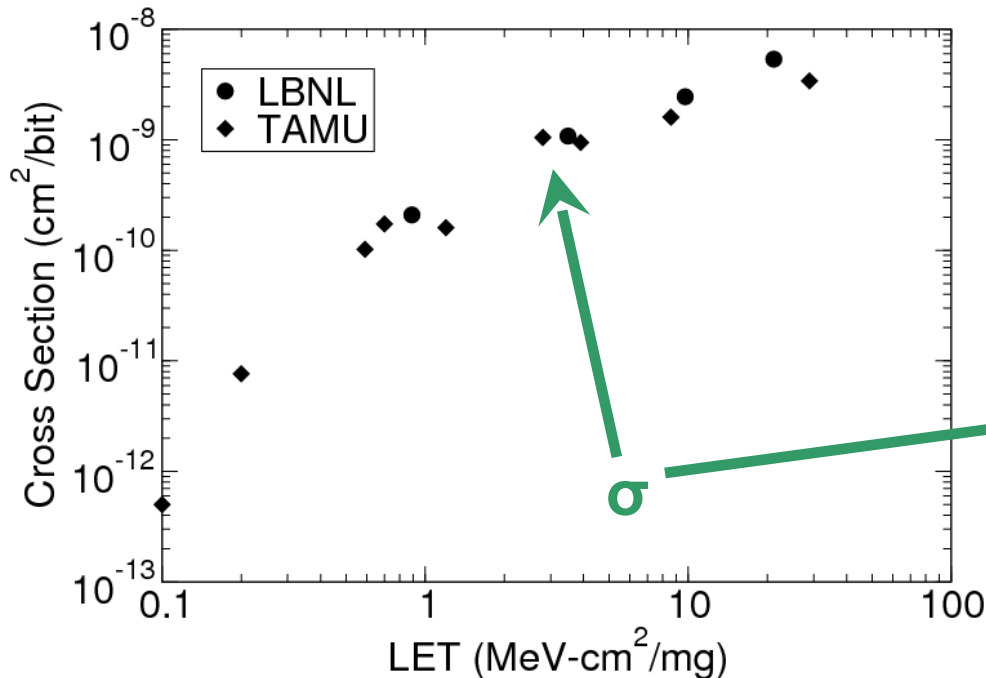


# Modeling from Heavy Ion Data



Vanderbilt Engineering

- Assume that single bit cross sections correspond to physical device area
- Low-LET heavy ion cross sections used to define sensitive areas
  - Single, well-known stopping power
- Allow nuclear transport codes and calibrated model to predict low energy proton response





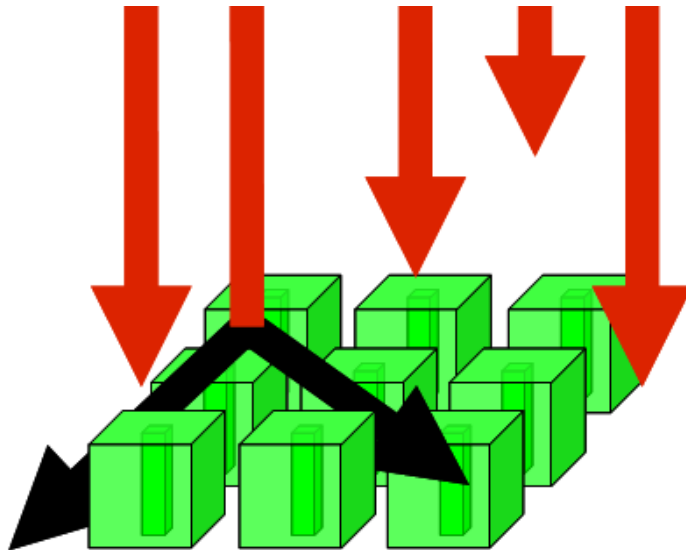


# Monte Carlo Simulation

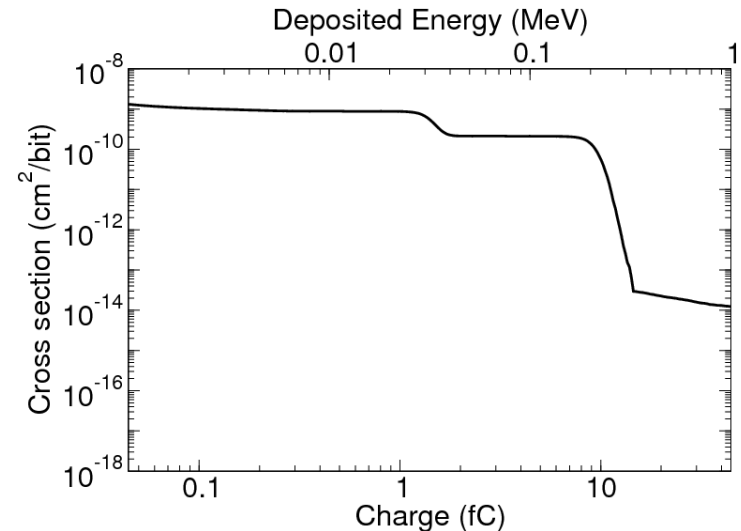
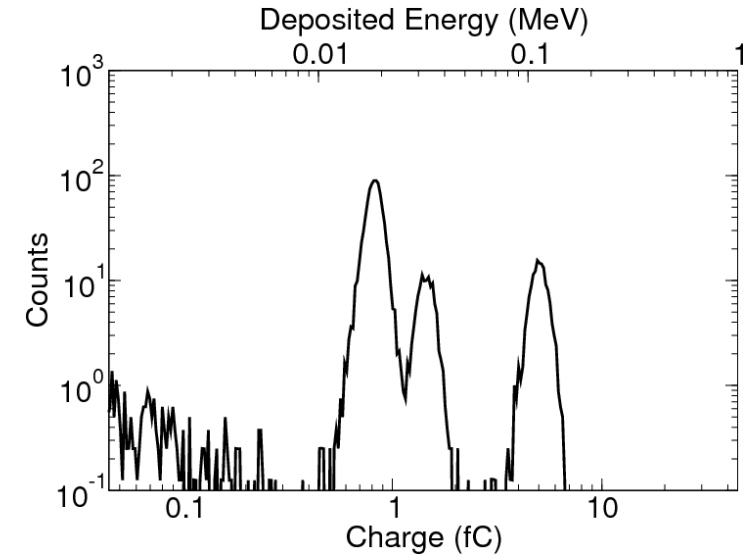


Vanderbilt Engineering

- **Vanderbilt's MRED code randomizes protons over array of devices**
- **Captures energy deposition from direct ionization, recoils, and nuclear reactions and weights to collected charge**



$$Q_{coll,j} = \frac{pC}{22.5MeV} \sum_{k=1}^{N_j} a_{k,j} E_{dep,j,k}$$

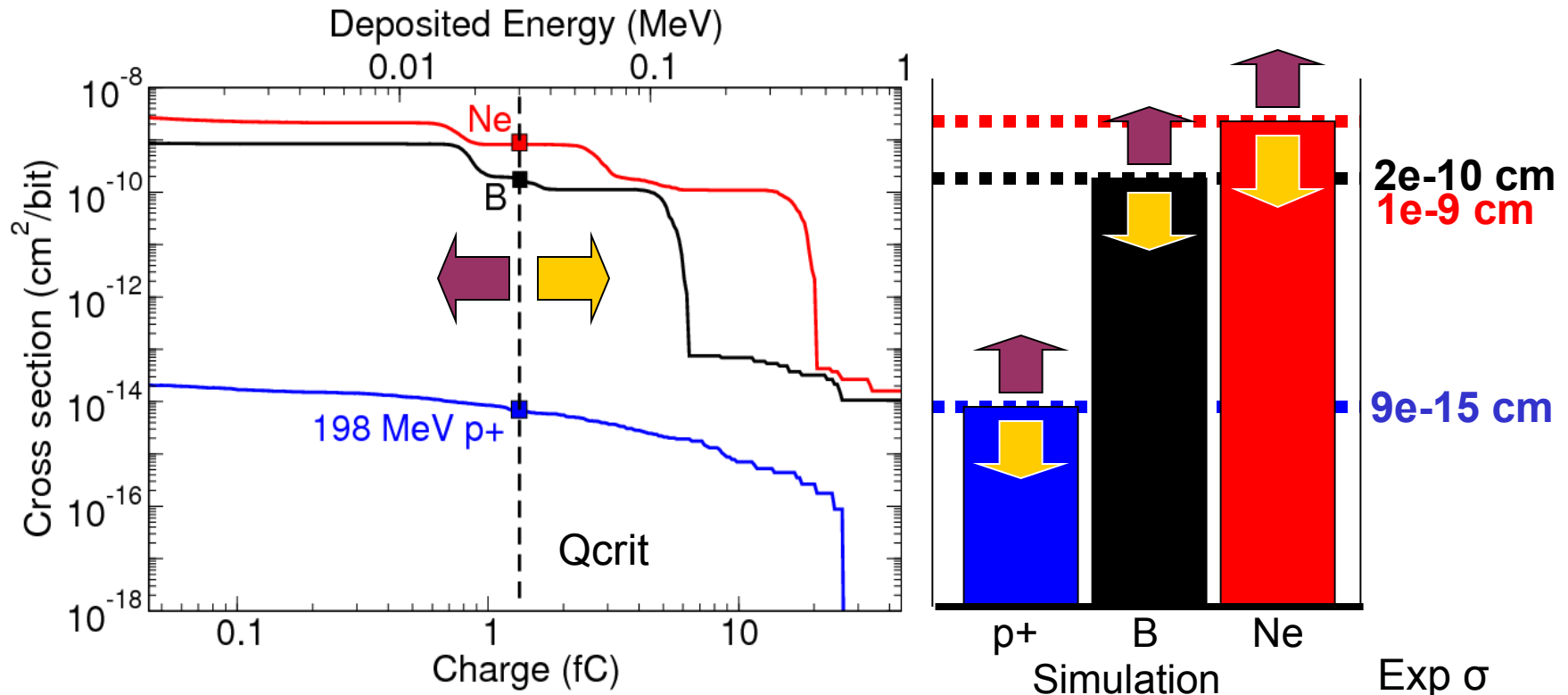




# Qcrit Parameter Estimation



Vanderbilt Engineering



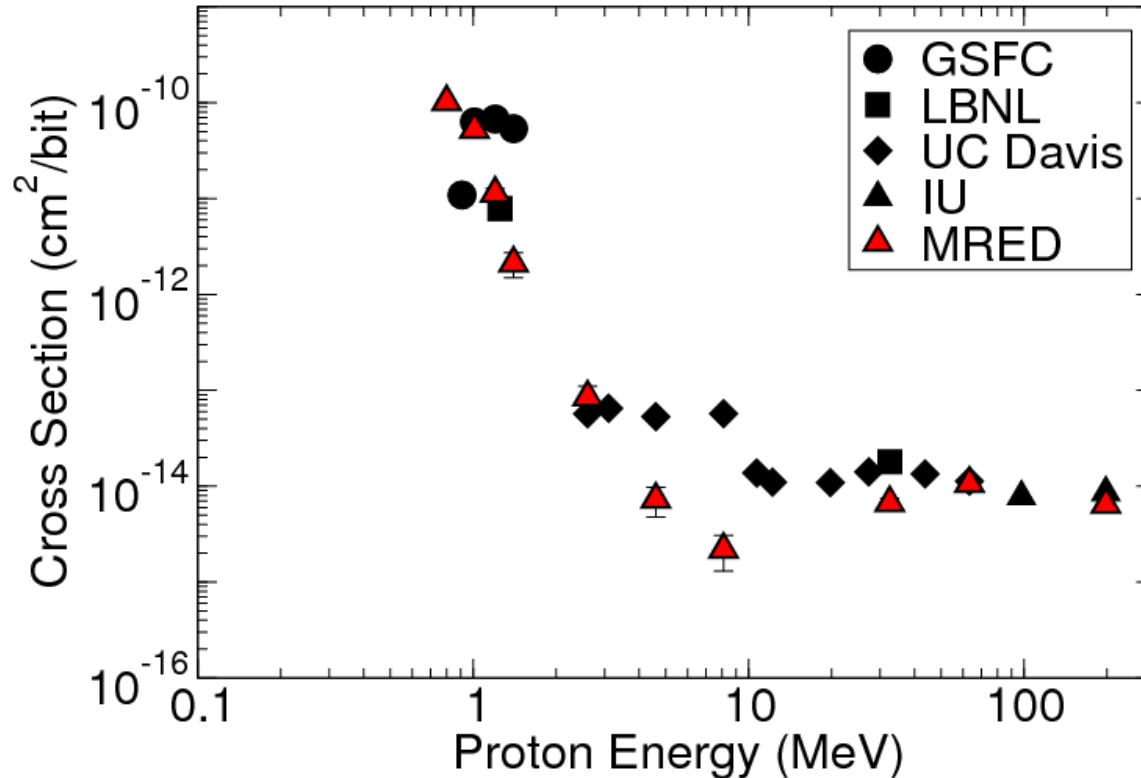
- MRED produces SEU cross section as a function of collected charge
- Least-squares fit to heavy ion and high energy proton SEU cross sections is used to determine the circuit critical charge (Qcrit)
- Fit estimates a Qcrit of 1.3 fC, Spice simulations estimated 1.4fC



# Proton Predictions



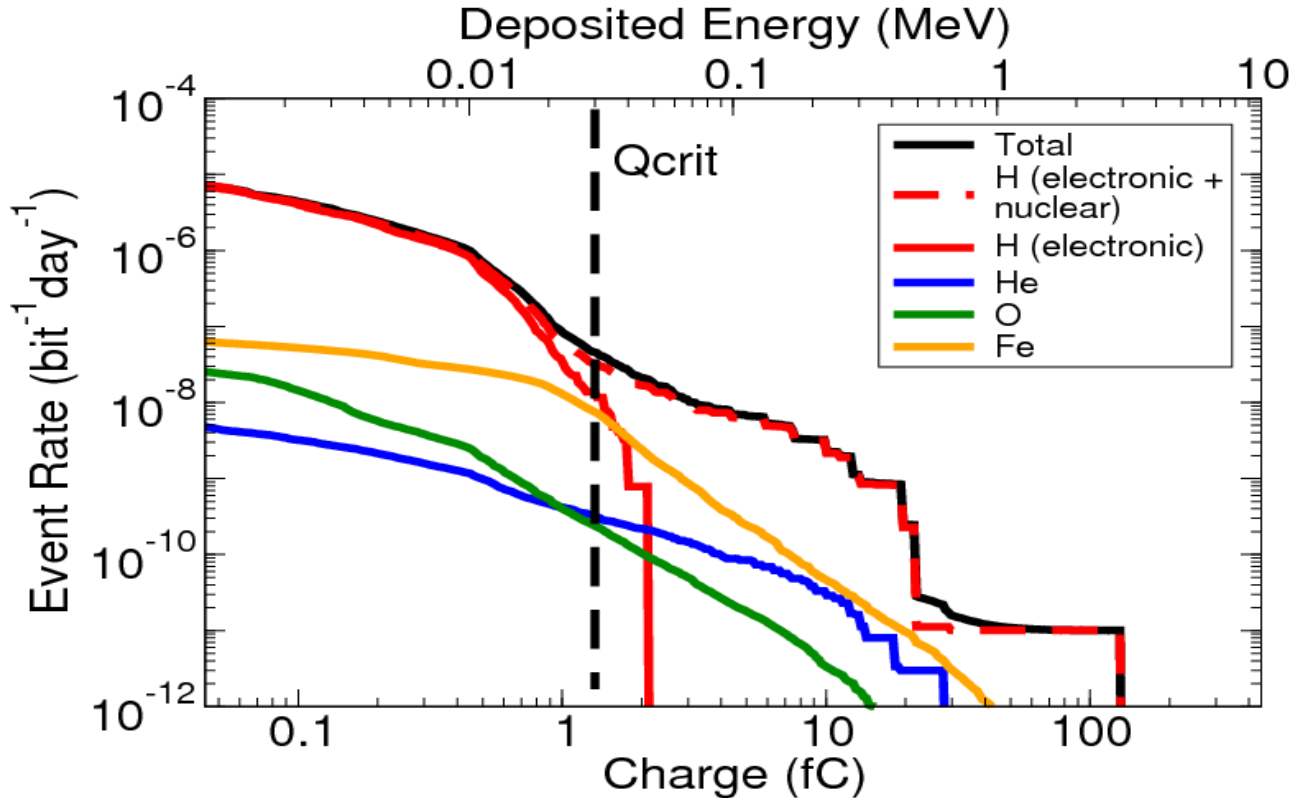
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- Estimation of critical charge from previous slide yields the SEU cross section over all proton energies
- Underprediction in 3 – 10 MeV range possibly due to spread in beam energy



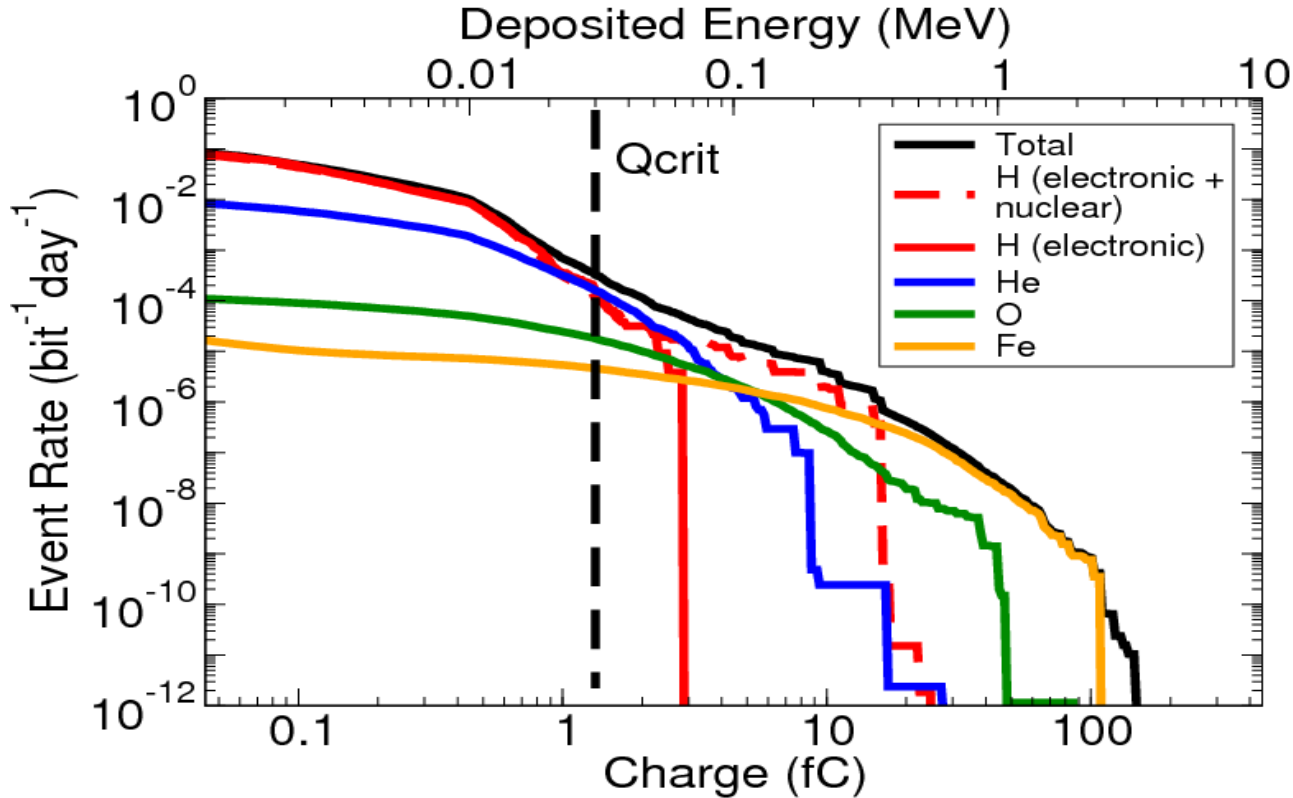
# Contribution of Protons to Error Rate in ISS Orbit



- Applying isotropic space environment (AP8MIN, magquiet, solar min, 100 mils Al) to sensitive volume model reveals error rate as function of species and critical energy
  - Direct ionization is becoming the dominant upset mechanism for protons



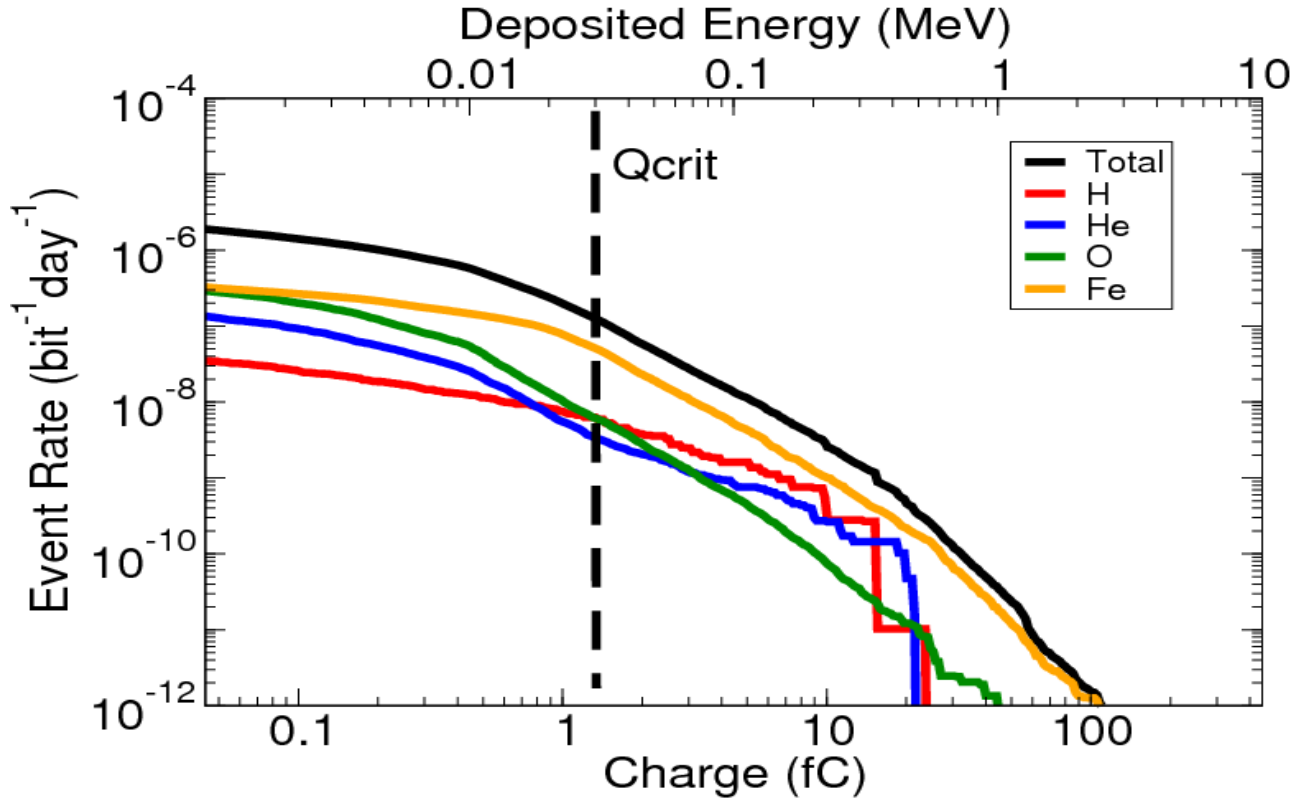
# Contribution of Protons to Error Rate in GEO (Worst Day)



- **Worst Day environment shows large contributions to error rate from both protons and alpha particles**
  - Need to assess impact on reliability



# Contributions of Species to Error Rate in GEO (Solar Min)



- Iron dominates rate in the GEO solar min with 100 mils Al shielding environment
  - Proton flux too low to be an issue
- Low-LET particles will drive the rate as technology scaling continues



# Low Energy Proton Testing Implications



- **Details regarding materials upstream from the sensitive DUT regions are important**
  - Kapton/aramica windows, degrader foils, air gap, substrate or BEOL thickness, PCBs, package lids, etc
- **Tune the primary beam energy as much as is feasible to achieve lower particle energy**
  - Don't forget straggle (range AND energy)
- **Nearly unavoidable systematic error in proton energy at DUT plane**
- **Best used to show device sensitivity**



# Summary



- **< 2 femtoCoulombs of collected charge can induce errors in modern circuits**
  - direct ionization from protons, alpha particles, and low-LET heavy ions
- **The high abundance of these particles makes a detailed understanding of the SEU response critical**
- **Tests with low-energy protons are difficult to interpret**
  - Use of data can be difficult due to uncertainty and variability in  $dE/dx$ , limited range of particles, transport through back-end-of-line
  - High-energy, low-LET ions may be better suited for characterization in this regime
- **MRED models fit experimental data and can be used with higher confidence for predicting SEU rates in arbitrary space environments**